

# The Effects of Shot Peening on the Surface Characteristics of 35NCD16 Alloy Steel

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## Abstract

Shot-peening is a surface mechanical treatment, widely used to treat metallic parts in the aerospace and automotive industries. This mechanical surface treatment should not be confused with other common applications of other peening treatments, oriented towards cleaning, preparation or surface finishing. Shot peening is a mechanical treatment used to improve the service life of metallic components, especially when they are submitted to cyclic loads. The effect of shot peening on the surface work hardening and residual stress profile of a 35NCD16 steel was studied in this work.

## Keywords

shot peening, surface treatments, hardening, residual stress, micro hardness, FWHM

## 1 Introduction

The improvement of the local properties of the steel by a surface treatment can be done by thermal, thermochemical or even mechanical means. These treatments increase the surface hardness, by metallurgical transformations or by mechanical hardening. The increase in hardness generally ensures good resistance to wear and also to fatigue due to repeated loading [1, 2].

Most of these treatments also introduce compressive stresses in the surface layers of the component due to deformation heterogeneities induced within the work-piece (plastic deformations, volumetric expansions due to changes in temperature or to transformation of certain metallurgical phases).

Shot peening is one of these surface treatments. Referring to the literature dealing with shot peening applied to steels and its effects under different cyclic loadings [3–12], it is nowadays clear that shot peening is a very effective and cost efficient final treatment to increase the fatigue strength of steels. This surface treatment consists in bombarding the surface of the component with a stream of small, high hardness spheres, called shots. These impacts plastically

deform the surface layer (increasing local hardness), while the underneath region remains elastic, given rise to a compressive residual stress field [13–20], as it is shown in Fig. 1.

The most important parameters of the residual stress field are the surface compressive residual stress, the maximum compressive residual stress and the total depth of the compressive stress layer and these values depend on the shot peening parameters, shot size and density, Almen

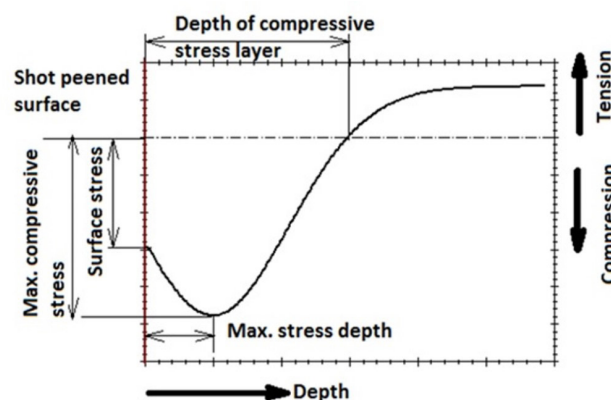


Fig. 1 Effects of shot peening

intensity, coverage [21, 22]. In order to find the optimum shot peening conditions for particular steel many variables are involved, as steel grade and strength, surface roughness, applied loads [23].

The existence of a residual compressive stress field significantly improves the fatigue behavior of any particular steel component, especially when it is submitted to cyclic bending loads, mainly due to two reasons. First, due to surface work hardening, and secondly, because fatigue cracks only open and grow under tensile loads, so that compressive stresses produces a crack closure effect.

The intensity of a shot peening treatment is measured by the so-called Almen intensity, which is an indirect measurement of the energy of the shot stream when it impacts into the target [24]. Moreover, coverage is defined as the ratio of the area covered by the shot impacts (dimples) to the complete surface of the treated sample, expressed as percentage. The degree of coverage does not increase linearly as a function of the peening time, the approximation to full coverage being exponential, so that full coverage is conventionally assessed when the target has an impacted area of 98 % [25–29].

Under this assumption, a degree of coverage of 200 % corresponds to a peening treatment twice as long as the full coverage treatment [30, 31]. Incomplete coverage of the surface produces a heterogeneous surface compressive stress field, resulting in earlier nucleation of fatigue cracks, while an overlong peening time gives rise to excessive surface plastic deformation, which can produce surface defects, worsening the fatigue performance of the material [31]. An experimental investigation about this material had been done by Bignonnet and Charkaluk [32]. The results of the study show that this design tool on fatigue developed by the Laboratory of Mechanical Systems Engineering (LASMIS) [33] is able to take into account different loading parameters.

The aim of this paper is to present a systematic study of the effects of conventional shot peening on the surface hardening and residual stress profiles of a 35NCD16 alloy steel.

## 2 Fundamental equations for the shot peening

The residual stresses induced by shot peening are expressed by following the Li et al.'s mechanical approach [34].

The classical Hertz results are used to model the elastic stress field created by the impact. The Hertzian stresses are derived for a position directly below the indenter. The reason for choosing a location exactly below the impacting shot is to simplify the problem. As in [35] state of zero

shear stress exists there, thus the stresses are principal and the Hertzian elastic stress tensor can be written as follows:

$$T^e = \begin{bmatrix} \sigma_{11}^e & & \\ & \sigma_{22}^e & \\ & & \sigma_{33}^e \end{bmatrix}, \quad (1)$$

where the subscripts 1, 2 and 3 represents the axis  $x_1$ ,  $x_2$  and  $x_3$ , respectively. Fig. 2 gives an idealized illustration of a single shot affecting a surface along with expressions for the indentation radius,  $a_p$ , and maximum elastic pressure,  $p$ , derived from Hertzian contact theory.

The Hertzian stresses are determined for a position straightforwardly underneath the indenter. The reason for choosing an area exactly below the impacting shot is to simplify the problem.

The corresponding mean stress and strain are:

$$\sigma_m^e = \frac{1}{3}(\sigma_{11}^e + \sigma_{22}^e + \sigma_{33}^e), \quad (2)$$

$$\epsilon_m^e = \frac{1}{3}(\epsilon_{11}^e + \epsilon_{22}^e + \epsilon_{33}^e). \quad (3)$$

The elastic deviatoric strain components are found by subtracting the mean strains from the strain tensor. The principal components of this tensor are derived as:

$$e_{11}^e = \epsilon_{11}^e - \epsilon_m^e = \frac{1}{3}(1+\nu)\epsilon_i^e, \quad (4)$$

$$e_{22}^e = \epsilon_{22}^e - \epsilon_m^e = \frac{1}{3}(1+\nu)\epsilon_i^e, \quad (5)$$

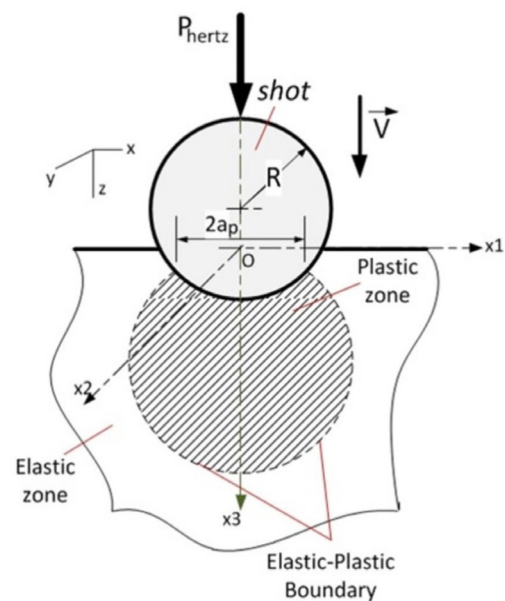


Fig. 2 Schematic of a rigid shot impacting a semi-infinite surface

$$e_{33}^e = \epsilon_{33}^e - \epsilon_m^e = -2e_{11}^e = -\frac{2}{3}(1+\nu)\epsilon_i^e. \quad (6)$$

The principal components of the deviatoric stress tensor in the target material can be expressed as:

$$S_{11}^e = \sigma_{11}^e - \sigma_m^e = \frac{1}{3}\sigma_i^e, \quad (7)$$

$$S_{22}^e = \sigma_{22}^e - \sigma_m^e = \frac{1}{3}\sigma_i^e, \quad (8)$$

$$S_{33}^e = \sigma_{33}^e - \sigma_m^e = -2S_{11}^e = -\frac{2}{3}\sigma_i^e. \quad (9)$$

As in numerous elastic-plastic investigations, the elastic-plastic constitutive relationship associated with the target material can be simplified to a multi-linear one.

A simplified relation [34] will be executed here. By adopting a modifying coefficient,  $\alpha$ , the elastic-plastic strain  $\epsilon_i^p$  at any depth  $x_3$  is calculated from the elastic strain  $\epsilon_i^e$ , at the corresponding depth  $x_3$ :

$$\epsilon_i^p = \begin{cases} -\epsilon_i^e & \text{for } \epsilon_i^e \leq \epsilon_s \\ \epsilon_s + \alpha(\epsilon_i^e - \epsilon_s) & \text{for } \epsilon_i^e > \epsilon_s \end{cases}. \quad (10)$$

The linearity coefficient  $\alpha$  is defined as the ratio of the plastic and elastic indentation radius [34, 36],

$$\alpha = \frac{a_p}{a_e} \quad (11)$$

and  $\epsilon_s$  is the strain at yielding.

According to the Hertzian contact theory, when the elastic compression is at its maximum, the radius of the elastic contact circle between the shot and the semi-finite body, is expressed as:

$$a_e = R \left( \frac{5}{2} \pi \kappa \rho \frac{V^2}{E_0} \right)^{\frac{1}{5}}, \quad (12)$$

where  $V$  is the shot velocity,  $R$  is the shot radius,  $\nu$  is the target Poisson ratio,  $\nu_s$  is the shot Poisson ratio,  $\rho$  is the shot density and  $\kappa$  is an efficiency coefficient based upon elastic rebound of the shot [37].

The maximum plastic indentation radius of a dent produced by the same shot on an elastic-plastic target under the same velocity [35] is

$$a_p = R \left( \frac{8\rho V^2}{9\sigma_s} \right)^{\frac{1}{4}}, \quad (13)$$

where  $\sigma_s$  is the yield strength.

According to [34], the equivalent stress,  $\sigma_i^p$ , is solved by assuming a multi-linear relationship with the equivalent plastic strain:

$$\sigma_i^p = \begin{cases} -\sigma_i^e & \text{for } \epsilon_i^e < \epsilon_s \\ \epsilon_s + \eta(\epsilon_i^p - \epsilon_s) & \text{for } \epsilon_s \leq \epsilon_i^p < \epsilon_b \\ \sigma_b & \text{for } \epsilon_i^p \geq \epsilon_b \end{cases}. \quad (14)$$

The quantities  $\sigma_s$ ,  $\sigma_b$ , and  $\eta$  are the yield stress, ultimate tensile stress and linear strain hardening parameter respectively (Fig. 3).

A great simplification is made by assuming the deviatoric plastic strains take on a similar form as the deviatoric elastic strains. In order to obtain the stress deviators, that are the components of the stress tensor that effectively contribute to the plastic deformation, the strain deviators have to be obtained. Due to the asymmetry of loading the elastic-plastic strain deviators,  $e_{ij}^p$ , are expressed as:

$$e_{11}^p = \frac{1}{3}(1+\nu)\epsilon_i^p, \quad (15)$$

$$e_{22}^p = \frac{1}{3}(1+\nu)\epsilon_i^p, \quad (16)$$

$$e_{33}^p = -2e_{11}^p = -\frac{2}{3}(1+\nu)\epsilon_i^p. \quad (17)$$

The next goal is to derive the elastic-plastic deviatoric stresses. Based on the Ilyushin's Elastic-plastic

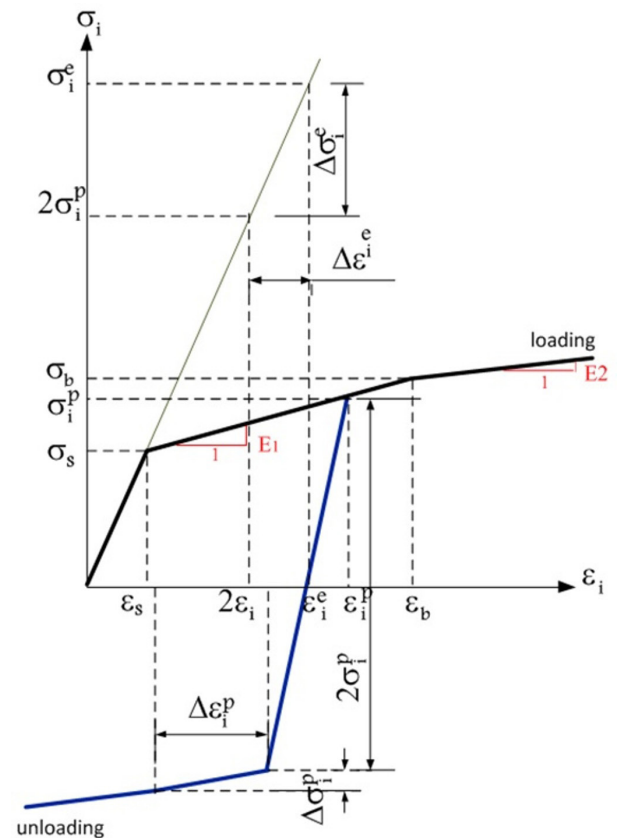


Fig. 3 Schematic diagram for calculating stresses after unloading

theory [38, 39], the strain deviators  $e_{ij}^p$  ( $j = x1, x2, x3$ ) and the elastic-plastic strain  $\epsilon_i^p$  were taken into Eq. (18), then the stress deviators of elastic-plastic model  $S_{ij}^p$  ( $j = x1, x2, x3$ ) can be obtained, as shown in Eqs. (19–21):

$$S_{ij}^p \epsilon_i^p = \frac{\sigma_i^p}{1+\nu} e_{ij}^p \quad (18)$$

$$S_{11}^p = \frac{\sigma_1^p}{(1+\nu)\epsilon_1^p} e_{11}^p = \frac{\sigma_1^p}{3} \quad (19)$$

$$S_{22}^p = \frac{\sigma_2^p}{(1+\nu)\epsilon_2^p} e_{22}^p = \frac{\sigma_2^p}{3} \quad (20)$$

$$S_{33}^p = \frac{\sigma_3^p}{(1+\nu)\epsilon_3^p} e_{33}^p = -2 \frac{\sigma_3^p}{3}. \quad (21)$$

Finally, the residual stresses resultant from the shot peening process is obtained after the elastic–plastic analysis of the forces that result from the Hertzian elastic contact. Assuming that the amount of deformation is small, unloading is an elastic process until reversed yielding starts and that hydrostatic stress does not introduce plastic deformation, the essential formula for residual stress calculation is [38, 40]:

$$\sigma_{ij}^r = S_{ij}^p - S_{ij}^e. \quad (22)$$

The shot peened material (target material) is assumed to be isotropic. Therefore, the residual stresses can be derived from the following relation [39]:

$$\sigma_{ij}^r = \begin{cases} 0 & \text{for } \sigma_i^e \leq \sigma_s \\ S_{ij}^p - S_{ij}^e & \text{for } \sigma_s \leq \sigma_i^e < 2\sigma_i^p \end{cases}, \quad (23)$$

which illustrates that there is no residual stress in elastic loading stage. Thus, the residual stresses can be expressed as [20]:

$$\begin{aligned} \sigma_{11}^r &= \frac{1}{3}(\sigma_1^p - \sigma_1^e) \\ \sigma_{22}^r &= \frac{1}{3}(\sigma_2^p - \sigma_2^e) \quad \text{for } \sigma_s \leq \sigma_i^e \leq 2\sigma_i^p. \\ \sigma_{33}^r &= -2\sigma_{11}^r \end{aligned} \quad (24)$$

As mentioned by [38], the final residual stresses for 100 % coverage ( $\sigma_{11}^R$ ) can be obtained by:

$$\sigma_{11}^R = \sigma_{22}^R = \frac{1+\nu}{1+\nu} \sigma_{11}^r. \quad (25)$$

As shown in Fig. 3, the reverse yielding and hardening occurs when  $\Delta\sigma_i^e = \sigma_i^e - 2\sigma_i^p \geq 0$ . The stress  $2\sigma_i^p$  is elastically unloaded first.

### 3 Material and experimental procedure

#### 3.1 Material

A 35NCD16 steel (36NiCrMo16, Europe EN10083), widely used in the automotive and aeronautical industries for the manufacture of gears, shafts and other structural components was considered in this study. The nominal chemical composition of the steel and its mechanical properties are respectively shown in Tables 1 and 2. Fig. 4 shows the microstructure of the steel, chemically etched with Nital-2 %.

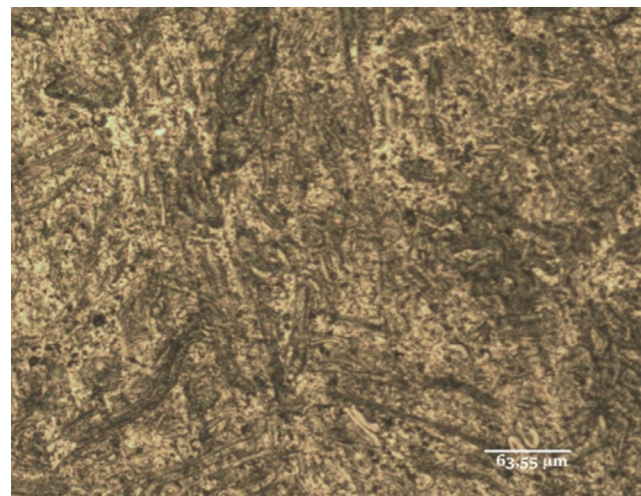
Shot peening was performed by means of a direct compressed air machine (GUYSON Euroblast 4 PF) using conditioned cut wire shots with rounded off edges (CW, 670–730 HV, with an average diameter of 0.4 mm and 0.6 mm) and 4 bar [41]. Table 3 shows shot diameter, Almen intensity [27] and surface coverage [42] applied in the different conventional shot peening treatments (CSP1, CSP2 and CSP3).

**Table 1** Nominal chemical composition of 35NCD16 steel (in wt.%)

C	Mn	Si	Ni
0.32–0.39	0.5–0.8	0.10–0.40	3.60–4.10
Cr	Mo	S	
1.60–2.0	0.25–0.45	0.025	

**Table 2** Mechanical properties of 35NCD16 steel (elastic modulus, yield strength, ultimate tensile strength, elongation and reduction of area)

Material	$V_{\text{test}}$ (mm/min)	$E$ (GPa)	$\sigma_y$ (MPa)
35NCD16	3	202	486
	(MPa)	e [%]	RA [%]
	622	27.6	70



**Fig. 4** Microstructure of 35NCD 16 steel



**Table 3** Shot peening treatments

Treatment	Shot diameter (mm)	Almen intensity	Pressure (MPa)	Coverage (%)
Not shot peened (NSP)	-	-	-	-
Conventionally shot peened 1 (CSP1)	0.4 mm	10 A	0.4	200
Conventionally shot peened 2 (CSP2)	0.6 mm	10 A	0.2	200
Conventionally shot peened 3 (CSP3)	0.6 mm	10 A	0.4	200

### 3.2 Hardness measurements

After the shot peening treatments, the samples were transversally cut, embedded in a cold-mounting resin, ground and polished for the measurement of hardness. Vickers microhardness indentations with a load of 200 g using a dwell time of 15 s were performed from the treated surface until attaining a depth at which the initial hardness was not modified by the treatment. These tests were performed using a Buehler Micromet 2100 microhardness tester, in accordance to the ASTM E384 standard.

### 3.3 Analysis of residual stress profiles

The X-ray diffraction technique was used to measure the residual stress field produced by the shot peening treatment. These measurements were carried out in depth, step by step, removing a very thin layer of material by means of electro-polishing, in order to obtain the in-depth residual stress profile until compressive stresses disappeared. A solution of acetic acid (94 %) and perchloric acid (6 %) was used for electro-polishing. The results of the in-depth residual stress measurements were corrected using the method described by Moore and Evans [43] in order to account for the removed material. Residual stress measurements were performed in an X-ray diffractometer (Stresstech XSTRESS 3000 G3R), by means of the  $\sin^2\psi$  method [4, 24, 43, 44]. A Cr- $K\alpha$  X-ray source with a wavelength of 0.2291 nm ( $K\alpha = 0.2291$  nm) was generated under a voltage 30 kV and a current of 6.7 mA. Measurements were taken on the (211) diffraction plane of the ferrite/martensite phase, which was recorded at a  $2\theta$  angle of approximately  $156^\circ$ , and the elastic constant of the selected diffraction plane,  $E/(1 + \nu)$ , was 168.9 MPa [5, 25, 45, 46]. Diffraction data were determined in three different  $\phi$  directions on the sample plane:  $-45^\circ$ ,  $0^\circ$  and  $+45^\circ$ , subsequently calculating the average result. The longitudinal direction

of the samples was designated as  $\phi = 0$ . Nine tilt angles ( $\psi$ ) between  $-45^\circ$ ,  $0^\circ$  and  $+45^\circ$ , a 40-second exposure time for each single measurement and a 2 mm diameter collimator were also used. The diffraction peaks were processed via a Pseudo-Voigt adjustment and the background noise was minimized by means of a parabolic expression. The summary of measurement settings is presented in Table 4. In the course of the diffraction measurements, the FWHM parameter, that is the full width of the diffraction peak at its half maximum, was also recorded, this parameter being related with the material hardness.

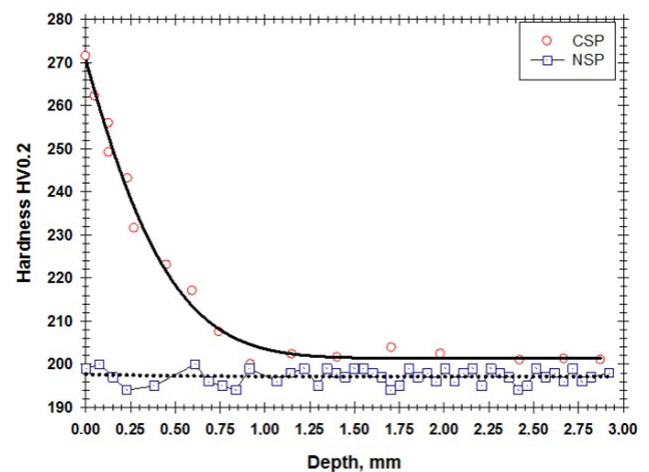
## 4 Results and discussion

### 4.1 Microhardness

Fig. 5 shows the in-depth microhardness variation from the treated surface to the bulk material of the shot peened specimens (CSP 1, 2 or 3). It is seen that surface hardness significantly increased up to 25 % in comparison with NSP specimen due to work hardening.

**Table 4** Experimental parameters employed in the X-ray diffraction analysis.

Wavelength $K\alpha$ (Cr)	0.2291 nm	Filter	Vanadium
Exposure time (s)	40	$\emptyset$ collimator (mm)	2
Tilt $\psi$ ( $^\circ$ )	9 points between $-45/+45$	Rotation angle, $\phi$ ( $^\circ$ )	$-45$ , 0 and 45
Background	Parabolic	Fit	Pseudo-Voigt
Measuring mode	$\chi$ -modified	Diffraction angle	$156^\circ$
Miller indices (hkl)	(211)	Elastic constant, $E/(1 + \nu)$ (GPa)	168.9 $\pm$ 2.8



**Fig. 5** Hardness profiles

## 4.2 Residual stress profiles

Conventional shot peened specimens show high surface compressive stresses between 290 and 340 MPa (0.6–0.7 y), maximum compressive stresses of 350–390 MPa (0.7–0.8 y) and depth submitted to compressive residual stresses varied between 0.25 to 0.27 mm (Fig. 6). It is also noted the slightly lower residual stress and depth submitted to RS of the treatment applied with the smallest shots (CSP1, CW0.4). The residual stress profile of the non-treated specimens is also recorded in the same figure: any significant residual stress is present, except in a very narrow surface region. The small compressive residual stress measured in this small region is unduly due to machining.

## 4.3 Estimation of work-hardened layer thickness

As it is observed in Fig. 7, the FWHM parameter significantly increases in the near-to-surface region and differences between CSP2 and CSP3 shot peening treatments are barely significant. It is also worth noting that depth

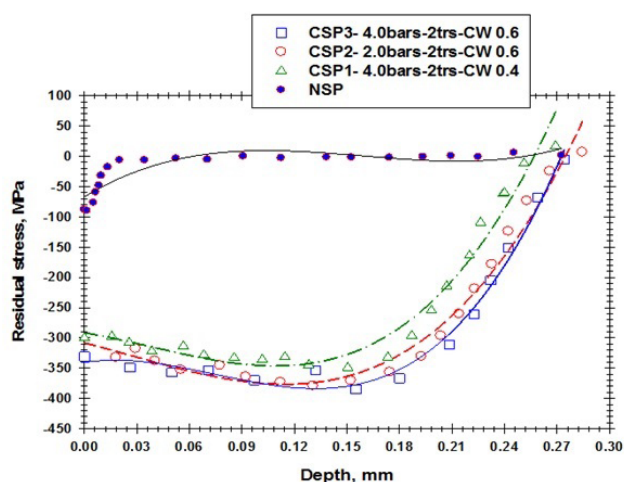


Fig. 6 Residual stress profiles

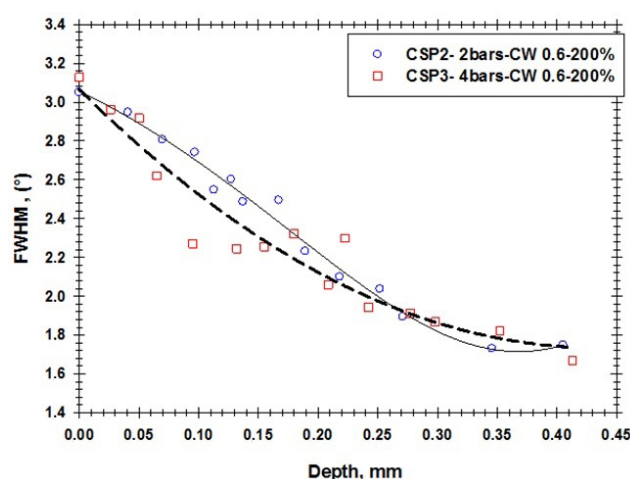


Fig. 7 Full Width at Half Maximum profiles

of the region in which the FWHM parameter increased in relation of the value in the non-affected steel (about 0.35 mm in Fig. 5) is quite similar to the depth submitted to compressive residual stresses (0.25–0.27 in Fig. 4). The depth of the compressed layer approximately corresponds to the depth of the work hardened layer.

Measurements of FWHM parameter were plotted versus the Vickers microhardness using results obtained from CSP2 and CSP3 treatments and a good linear regression was obtained (Fig. 8). It was then demonstrated the use of FWHM as a hardening parameter.

## 5 Conclusions

- Conventional shot peening treatments under a 10 A intensity give rise to high surface (0.6–0.7 y), and maximum (0.7–0.8 y) compressive stresses, along with a compressive layer depth of 0.25–0.27 mm.
- For the same Almen intensity, slightly lower residual stresses and compressive stress depth were produced in the SP treatment applied using the smallest shots.
- An almost linear correlation was obtained between FWHM parameter and hardness, demonstrating the appropriate use of FWHM as a hardening parameter at least in shot peening applications.
- Shot peening treatments also induce an increase in the hardness of the surface region, due to work hardening, which can be easily and effectively quantified by means of the FWHM parameter.

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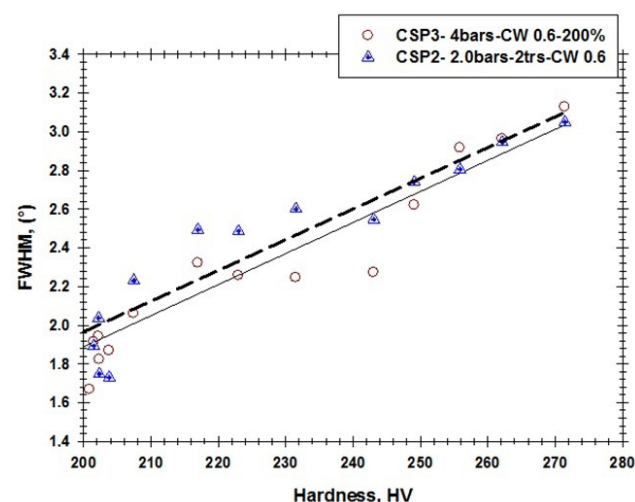


Fig. 8 FWHM parameter versus Vickers microhardness

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